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- 1. THE FOLLOWING INFORMATION REPRESENTS CORRECTIVE ACTION
 TO CLEAR DISCREPANCIES ON Ø58 WHICH CAUSED THE AIR ABORT ON
 25 SEP 68.
- A. A.C. GENERATOR FAILURE. CAUSED BY DECLUTCHING OF THE CSD UNIT. NEW CSD UNIT INSTALLED; DURING GROUND RUN A.C. SYSTEM APPEARS NORMAL. AS THIS WAS THE SECOND INSTANCE OF CSD FAILURE IN THE PAST FOUR FLIGHTS, THE CSD OIL TEMPERATURE CONTROL VALVE WAS REPLACED AS A PRECAUTIONARY MEASURE. DRY NITROGEN WAS BLOWN THROUGH THE CSD RADIATOR TO ASSURE THAT IT WAS NOT BLOCKED.
- B. AUTO PILOT INOPERATIVE DUE TO IRRADICT PITCH MOVEMENTS.

 SYSTEM WAS THOROUGHLY CHECKED AND A GROUND WIRE FROM THE PITCH

 RATE COCKPIT POT TO THE AUTO PILOT COMPUTER WAS DISCOVERED TO BE

 LOOSE AT THE COMPUTER PLUG. IT IS ASSUMED THAT BULKHEAD

 FLEXING OR IN-FLIGHT VIBRATIONS CAUSED THE INTERMITTANT PITCH

 DIFFICULTY; A TEMPORARY GROUNDING FIX WAS ACCOMPLISHED AND THE

 SYSTEM CHECKED OUT NORMALLY.
 - C. TACAN INOPERATIVE. DIFFICULTY WAS DISCOVERED TO

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WILL BE FLOWN ON 27 SEP 68.

3. PLAN NOW IS TO STANDBY AFTER LAUNCH UNTIL THE ARTICLE IS AT HALF WAY POINT. THEN LOAD THE C-130 WITH DEPARTURE OF C-130 AND C-118 ABOUT 1530L.

END OF MSG